Radar as a tool to study the Solar System

GBT/AO Single Dish Workshop September 16, 2021

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Outline

- What is a radar system and why do we use it for Solar System studies
- Introduction to Radar Systems and observing methods
- Results from planetary radar observations
- Introduction to Solar System radar data processing



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What is RADAR?



- RAdio Detection And Ranging
- Active remote sensing we control the signal!
- Maximize signal properties to enhance SNR
- Measure reflected radio waves (so targets must be reflective)



Image source: https://en.wikipedia.org/wiki/Klystron

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Klystron Amplifier



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- Range (distance) resolution as fine as ~5 m
- Relative velocity resolution as fine as ~ 10 mm/s

Image credit: https://ssd.jpl.nasa.gov/sbdb.cgi#top





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Image credit: Jon Giorgini (JPL)

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Recinto de Cupo

Image credit: Jon Giorgini (JPL)

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Image credit: Jon Giorgini (JPL)

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Recinto de Cupe

Image credit: Jon Giorgini (JPL)

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Arecibo, Goldstone, GBT

Arecibo, Goldstone, GBT

- Declination range: -46°–90° (86% of the whole sky)
- 100m antenna

The Radar Equation

Continuous Wave (CW)

Continuous Wave (CW) signals: This is when we transmit a continuous

- This is when we transmit a continuous signal with a **constant amplitude, phase, and frequency.**
- We cannot differentiate signals based on when they arrive (no range information).
 This only gives us information on the shift in frequency, or Doppler shift of the reflected signal, based on the motion of the target.

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Doppler frequency (Hz)

 Target's doppler bandwidth depends on size and spin state:

$$B = \frac{4\pi D \cos\phi}{\lambda_0 P_{spin}}$$

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Doppler equation: $f_{\rm o} = \left(1 \pm \frac{2\nu}{c}\right) f_{\rm e}$ $v = \frac{\pi D}{P}$ $f_{\rm e} = \frac{c}{\lambda}$ $B = \frac{2v}{c} f_{\rm e} - \left(-\frac{2v}{c} f_{\rm e}\right)$ $B = \frac{4\pi D}{\lambda P}$

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 Target's doppler bandwidth depends on size and spin state:

$$B = \frac{4\pi D \cos\phi}{\lambda_0 P_{spin}}$$

 RMS noise power depends on bandwidth, integration time, and system temperature:

$$V_{rms} = kT_{sys} \sqrt{\frac{B}{\tau}}$$
 k = Bo

k = Boltsmann's constant

- Target's doppler bandwidth depends on size and spin state:
- RMS noise power depends on bandwidth, integration time, and system temperature:
- Need SNR of ~6 for detection, but much higher SNR for imaging

 $N_{rms} = kT_{sys} \sqrt{\frac{B}{\tau}}$ k = Boltsmann's constant

 P_{rx}

$$SNR = \frac{P_{rx}}{N_{rms}}$$

$$=\frac{P_{tx}G_{tx}\sigma A_{eff}}{(4\pi)^2 r^4}$$

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$$P_{rx} = \begin{bmatrix} P_{tx} \end{bmatrix} \begin{bmatrix} G_{tx} \end{bmatrix} \begin{bmatrix} \frac{1}{4\pi r^2} \end{bmatrix} [\sigma] \begin{bmatrix} \frac{1}{4\pi r^2} \end{bmatrix} \begin{bmatrix} A_{eff} \end{bmatrix}^{s \text{ constant}}$$

 $SNR = \frac{P_{rx}}{N}$

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 P_{rx}

 $=\frac{P_{tx}G_{tx}\sigma A_{eff}}{\left(4\pi\right)^2 r^4}$

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$$N_{rms} = kT_{sys}\sqrt{\frac{B}{\tau}} \quad B = \frac{4\pi D\cos\phi}{\lambda_0 P_{spin}}$$

$$P_{rx} = \frac{P_{tx}G_{tx}\sigma A_{eff}}{(4\pi)^2 r^4} \quad SNR = \frac{P_{rx}}{N_{rms}}$$

SNR ~ (System factor)(Target factor) $\sqrt{\Delta t}$ System factor~ $\frac{P_{\text{tx}}G_{ant}A_{\text{eff}}}{T_{\text{sys}}\lambda^{1/2}}$ Target factor ~ $\frac{\hat{\sigma}D^{3/2}P^{1/2}}{r^4}$ $\hat{\sigma}$ = radar albedo

For every CW detection we learn:

- Astrometry: ~ 30 Hz offset in line-of-sight velocity of target
- Rotation rate: estimate from bandwidth and size

 $B = \frac{4\pi D \cos\phi}{\lambda_0 P_{spin}}$

• CPR: related to surface

roughness and composition

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Coded Waveforms

Coded (modulated) signals:

By adding some code, or time structure, to the transmitted signal, either **by varying the frequency, phase, or amplitude, we can determine ranges of targets**, or what distances from the telescope different portions of the signal were reflected from.

Convolve received signal with transmitted signal in **matched filter**.

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Binary Phase Coded Waveform

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One baud, defined as the length of time between phase shifts

 $\Delta r = \frac{c \bullet baud}{2}$

Code Bandwidth = 1/[baud][codelength]

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delay-Doppler Imaging

- Splitting signal into discrete range & frequency bins
- Requires high SNR

Time Delay: Distance

- Vertical axis
- Increasing downward
- Points at the top are closest to the observer

Doppler Frequency: Velocity

- Horizontal axis
- Increasing to the right
- Points on the right are approaching the observer

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3D – 2D ambiguity

Circles: points at the same distance (range) from observer Vertical Lines: points of constant Doppler shift Blue dots are blue shifted, red dots are red-shifted, green dots are not Doppler shifted

The two blue and red dots map to the same point in a radar image (north/ south ambiguity)

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Arecibo Observatory detects and characterizes ~100 near-Earth objects (NEOs) per year

Arecibo Observatory was detecting and characterizing ~100 near-Earth objects (NEOs) per year

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2017 YE5

Arecibo Observatory/NASA/NSF

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- Equal mass binary (only four known NEAs!)
- 7.5 m/pixel
- 20-24 hr period
- Albedo difference between bodies

2014 JO25

Arecibo Observatory/NASA/NSF

2014 JO25

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19 Apr 2017

- PHA
- Contact binary ~15 % of NEAs
- 7.5 m/pixel
- ~ 870 m
- 4.5 hr rotation period

3200 Phaethon

Didymos

- PHA
- Binary system
- 7.5 m/pixel
- •~760 m
- 2.3 hr rotation period
- Target of NASA's DART mission

- 2nd largest PHA known
- 75 m/pixel
- ~ 6 km
- 3.6 hr rotation period
- Surface features visible (crater or boulder?)
- Target of JAXA's DESTINY+ mission set to launch 2022

1998 OR2

- Observed April
 2020
- PHA
- 7.5 m/pixel
- ~2.3 km
- 4 hr rotation period
- Prominent crater visible in the " leading edge"

Arecibo Observatory/NASA/NSF

Science Return

- Precise astrometry
- Surface Properties
- Discover binary asteroids
- Shape and spin state
 - Recently confirmed by NASA's OSIRIS-REx mission to 101955 Bennu

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Mercury

- In the late 1960's Arecibo data determined the rotation rate to be 59 days, as opposed to 88 days previously accepted!
- Strong reflections from craters at the poles indicate

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- Thick cloudy atmosphere and intense pressure and temperature at surface makes optical/infrared remote sensing and landing missions difficult.
- Radar penetrates clouds, revealing surface and sub-surface features.

Venus

Determination of the core size, spin period, spin period variation, precession rate Spin state and moment of inertia of Venus (Margot et al. 2021)

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